FAILURE MODES EFFECTS ANALYSIS (FMEA) - CIL HARDWARE

NUMBER: 05-6BA-2206 -X

SUBSYSTEM NAME: EPD&C - LANDING GEAR CONTROL

REVISION: 2

02/13/90

PART DATA

PART NAME

VENDOR NAME

PART NUMBER

VENDOR NUMBER

LRU : FWD PCA 2

V070-763340

LRU

: FWD PCA 3

V070-763360

SRU : DIODE

JANTX1N1204RA

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:

DIODE, BLOCKING DIODE, RETURN BUS, FWD PCA NOS. 2 AND 3 (12A, 400V).

REFERENCE DESIGNATORS:

82V76A23CR14

83V76A24CR9

QUANTITY OF LIKE ITEMS: 2

TWO, ONE EA FPCA - 2 & 3

FUNCTION:

DIODES ISOLATE RETURN BUSES BETWEEN FWD PCA NO 2 AND FWD PCA NO 3. IN RETURN LINES OF REDUNDANT SHUTOFF VALVE. PROVIDES PARALLEL LINE TO RETURN BUS.

FAILURE MODES EFFECTS ANALYSIS FMEA - CIL FAILURE MODE

NUMBER: 05-6BA-2206-01

REVISION#:

4

07/01/99

SUBSYSTEM NAME: EPD&C - LANDING GEAR CONTROL

LRU: FWD PCA 2
ITEM NAME: DIODE

CRITICALITY OF THIS

FAILURE MODE: 1R3

FAILURE MODE:

OPEN, FAILS TO CONDUCT.

MISSION PHASE:

DO DE-ORBIT

VEHICLE/PAYLOAD/KIT EFFECT/VITY:

102 COLUMBIA

103 DISCOVERY

104 ATLANTIS

105 ENDEAVOUR

CAUSE:

STRUCTURAL FAILURE (MECHANICAL STRESS, VIBRATION), ELECTRICAL STRESS,

THERMAL STRESS, PROCESSING ANOMALY

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS

B) FAIL

C) PASS

PASS/FAIL RATIONALE:

A)

B)

FAILS "B" SCREEN BECAUSE DIODE FAILURE IS NOT FLIGHT DETECTABLE.

C)

- FAILURE EFFECTS -

(A) SUBSYSTEM:

FIRST FAILURE - LOSS OF CAPABILITY TO CONDUCT THROUGH ONE OF TWO RETURN PATHS. LOSS OF REDUNDANT RETURN TO THE REDUNDANT SHUTOFF VALVE.

(B) INTERFACING SUBSYSTEM(\$);

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FIRST FAILURE - LOSS OF CAPABILITY TO CONDUCT THROUGH ONE OF TWO RETURN PATHS. LOSS OF REDUNDANT RETURN TO THE REDUNDANT SHUTOFF VALVE.

(C) MISSION:

FIRST FAILURE - NO EFFECT

(D) CREW, VEHICLE, AND ELEMENT(S):

FIRST FAILURE - NO EFFECT

(E) FUNCTIONAL CRITICALITY EFFECTS:

PÓSSIBLE LOSS OF CREWIVEHICLE DUE TO LOSS OF ELECTRICAL POWER TO LANDING GEAR EXTEND VALVE PLUS LOSS OF ALL OTHER ASSOCIATED REDUNDANCY (LOSS OF RETRACT/CIRC VALVE, AND PYRO BACKUP) FOR EXTENDING THE LANDING GEARS.

-DISPOSITION RATIONALE-

(A) DESIGN:

REFER TO APPENDIX F, ITEM NO. 2 - DIODE

(B) TEST:

REFER TO APPENDIX F, ITEM NO. 2 - DIODE

GROUND TURNAROUND TEST

ANY TURNAROUND CHECKOUT TESTING IS ACCOMPLISHED IN ACCORDANCE WITH OMRSD.

(C) INSPECTION:

REFER TO APPENDIX F, ITEM NO. 2 - DIODE

(D) FAILURE HISTORY:

CÚRRENT DATA ON TEST FAILURES, FLIGHT FAILURES, UNEXPLAINED ANOMALIES, AND OTHER FAILURES EXPERIENCED DURING GROUND PROCESSING ACTIVITY CAN BE FOUND IN THE PRACA DATA BASE.

(E) OPERATIONAL USE:

NONE

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